

AIR LIFT I. DETERMINATION OF IN THE AIR LIFT TANK MOVING SOLID PARTICLE PARAMETERS

László Kovács – Sándor Váradi
 Professor emeritus– Associate Professor
 Budapest University of Technology and Economics
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Abstract

One of the most important vertical pneumatic conveying equipment is the air lift. The paper deals with the mathematical-physical model for describing the important physical parameters of the air lift. By applying the continuity and the momentum equations written on the control volume we have got differential equations. The equations has been solved using Runge – Kutta method. Diagrams - pressure, solid and air velocity, and concentration – obtained as a result of solutions gives a presentation on the basis of a given example.

Nomenclature

A_o [m²] particle cross section
 a_n [m/Pa] constant
 b [m] height of beginning of the pipe over distribution layer
 b_n [m/Pa] constant
 C_D [-] drag coefficient
 d [m] diameter
 d_o [m] diameter of solid particle
 F [N] drag force
 F_f [N] friction force
 f [-] friction factor
 G_1 [N] weight of a particle
 g [m/s²] acceleration of gravity
 H [m] height of the fluidized column in the air lift tank
 k_o [m/s] constant characteristic of the distribution layer
 m_1 [kg] mass of one particle
 \dot{m}_g [kg/s] mass flow of the gas flowing towards the nozzle over the distribution layer
 \dot{m}_{g1} [kg/s] mass flow of the conveying gas coming through the distribution layer
 \dot{m}_{gk} [kg/s] mass flow of the

$$\dot{m}_{g2} - \beta \dot{m}_{g1}$$

$$\rho$$
 [Pa]

$$\rho_o$$
 [Pa]

$$\rho_1$$
 [Pa]

$$\rho_k$$
 [Pa]

$$\rho_t$$
 [Pa]

$$\rho_j$$
 [Pa]

$$R$$
 [m]

$$R$$
 [J/kgK]

$$Re$$
 [-]

$$R_p$$
 [m]

$$R_t$$
 [m]

$$v$$
 [m/s]

$$v_{g1}$$
 [m/s]

$$v_j$$
 [m/s]

$$v_k$$
 [m/s]

$$v_m$$
 [m/s]

$$v_{mo} = \varphi v_m$$
 [m/s]

Greek letters

$$\alpha$$
 [degree]

$$\beta = \frac{\dot{m}_{g2}}{\dot{m}_{g1}}$$
 [-]

$$\varphi = \frac{v_{mo}}{v_m}$$

$$\mu$$
 [kg/ms]

$$\kappa$$
 [-]

$$\rho_b$$
 [kg/m³]

$$\rho_g$$
 [kg/m³]

$$\rho_{go}$$
 [kg/m³]

conveying gas flowing out of the nozzle

mass flow of the gas flowing across the tank

pressure

atmospheric pressure

pressure under the distribution layer

pressure of the gas at outlet of the nozzle

pressure of the fluidized tank over distribution layer in place $R = R_t$

pressure under the nozzle

radius

gas constant

Reynolds number

radius of the conveying pipe

radius of the air lift tank

velocity

velocity of the gas coming across the distribution layer at $R=R_1$

velocity under the nozzle

velocity at the outlet of nozzle

material velocity in the streaming layer

sedimentation velocity in the tank

angle

relative number

relative number

absolute viscosity of the air

ratio of specific heats

bulk density in the fluidized tank

gas density

gas density at

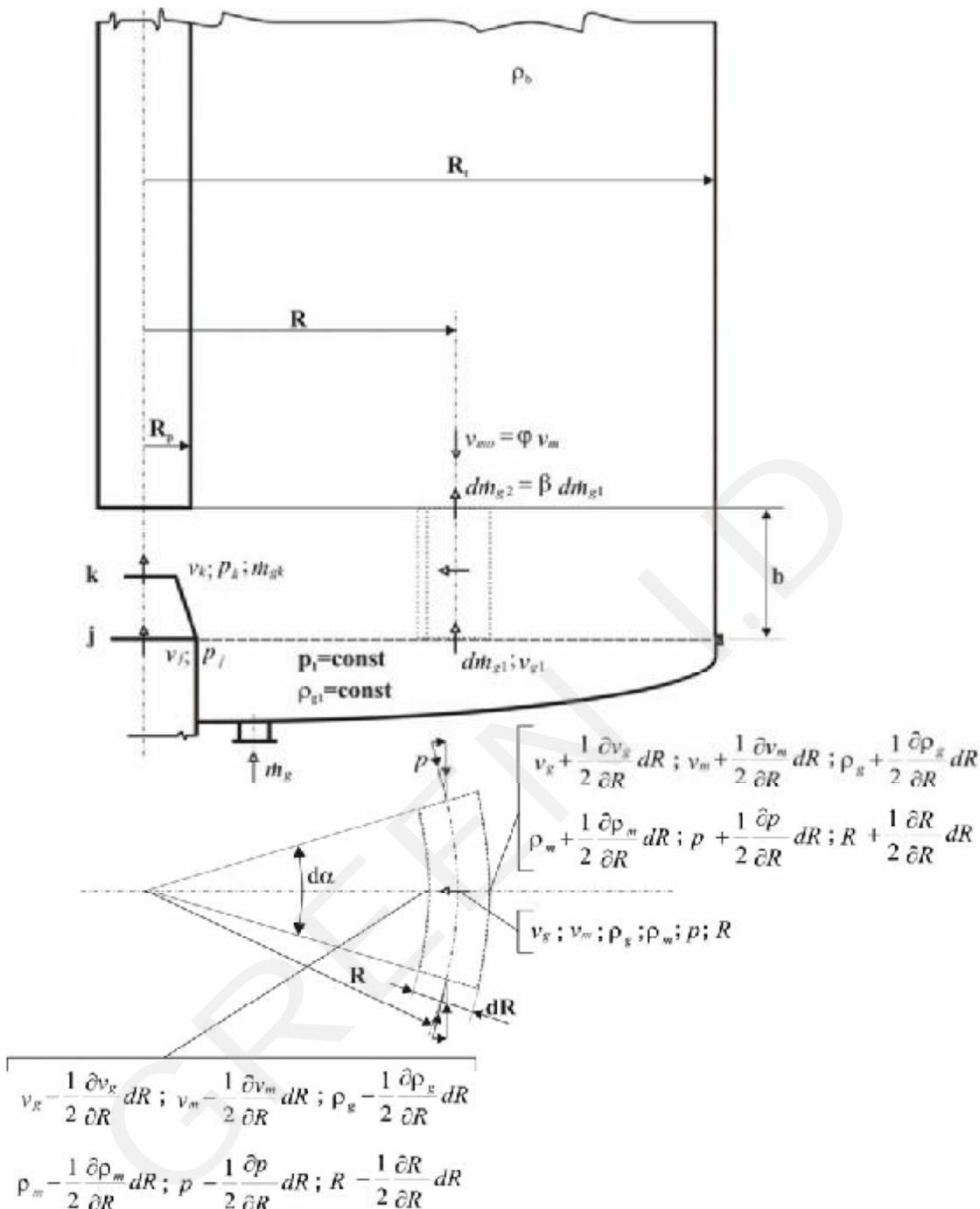


Fig. 2 Control volume in the radial flow on distribution layer

$$p_i - \rho_m gH \quad (1)$$

The air flowing out of the nozzle is considered to be frictionless, and the change in state adiabatic. The velocity of the gas leaving the nozzle is:

$$v_k = \left[2 \frac{\kappa}{\kappa - 1} RT_j \left(1 - \left(\frac{p_k}{p_j} \right)^{\frac{\kappa - 1}{\kappa}} \right) + v_j^2 \right]^{1/2} \quad (2)$$

Using velocity „ v_k ” taken from equation 2 and the known velocity „ v_j ” gives the following equation for pressure ratio „ p_k/p_j ”:

$$\frac{p_k}{p_j} = \left[1 - \frac{\kappa - 1}{\kappa} \frac{v_k^2 - v_j^2}{2RT_j} \right]^{\frac{\kappa}{\kappa - 1}} \quad (3)$$

Thus, pressure „ p_k ” can be determined of the value of „ p_j ”.

Assuming the variation in pressure levels to be parabolic, pressure distribution

over the distribution layer can be described by the following equation:

$$p = \frac{1}{2a_n} \left(-b_n + \sqrt{b_n^2 + 4a_n R} \right) \quad (4)$$

The constants of this equation can be calculated from pressure values „ p_t ” and „ p_k ”. These can be described as:

$$a_n = \frac{1}{p_k p_t} \frac{R_p p_k - R_t p_t}{p_t - p_k} \quad (5)$$

$$b_n = \frac{R_t - a_n p_k^2}{p_k} \quad (6)$$

Portion $dm_{g2} = \beta dm_{g1}$ of mass air flow „ dm_{g1} ” entering through an annulus surface element of the air distribution layer performs the fluidization of the column of „ H ” height in the tank and exits from the tank, while the balance marked „ dm_g ” entrains the particles as it flows towards the nozzle (the lower pressure place).

The mass flow of air flowing towards the nozzle is:

$$\begin{aligned} & dm_g - (1 - \beta) dm_{g1} - \\ &= (1 - \beta) \frac{p_1 - p}{k_o \rho_{g1}} \rho_{g1} R dR 2\pi = \\ &= (1 - \beta) \frac{2\pi}{k_o} \left[p_1 + \frac{b_n}{2a_n} - \frac{1}{2a_n} (b_n^2 + 4a_n R)^{1/2} \right] R dR \end{aligned} \quad (7)$$

The integration of this equation gives the following expression for the mass flow of air flowing towards the nozzle:

$$\begin{aligned} m_g = (1 - \beta) \frac{2\pi}{k_o} \left\{ \left(p_1 + \frac{b_n}{2a_n} \right) \frac{R_t - R}{2} - \right. \\ \left. - \frac{2}{480 a_n^3} \left[(12 a_n R_t - 2 b_n^2) (4 a_n R_t + b_n^2)^{3/2} - \right. \right. \\ \left. \left. - (12 a_n R - 2 b_n^2) (4 a_n R + b_n^2)^{3/2} \right] \right\} \quad (8) \end{aligned}$$

The calculation of the parameters of the solid material-air mixture flowing towards the nozzle will require the value of derivative „ dv_g/dR ” of the air velocity. Let

us set up the equation of continuity for this gas for the control volume marked in Figure 2 with the following assumptions taken into consideration:

- The tightening effect caused by the particles in the control volume will not be taken into account
- The change in the state of the gas is considered to be isometric.

$$\begin{aligned} R d\alpha dR \rho_{g1} \frac{p_1 - p}{k_o \rho_{g1}} + \\ + \left(R + \frac{dR}{2} \right) \left(\rho_g + \frac{d\rho_g}{2} \right) \left(v_g + \frac{dv_g}{2} \right) d\alpha b - \\ - \left(R - \frac{dR}{2} \right) \left(\rho_g - \frac{d\rho_g}{2} \right) \left(v_g - \frac{dv_g}{2} \right) d\alpha b - \\ - \beta R d\alpha dR \rho_{g1} \frac{p_1 - p}{k_o \rho_{g1}} = 0 \quad (9) \end{aligned}$$

After transformation and rearranging, as well as neglecting second-order members, the continuity equation can be written in the following form:

$$\frac{dv_g}{dR} = -(1 - \beta) \frac{(p_1 - p) p_o}{b k_o \rho_{g_o} p} - \frac{v_g}{R} - \frac{v_g}{p} \frac{dp}{dR} \quad (10)$$

(The gas mass flow calculated from the value obtained after the integration of equation 10 is identical with the value calculated with relationship 8.)

Figure 3 shows the radial distribution of gas mass flow $m_g(R)$ calculated with relationship 8, using the data of an example described later.

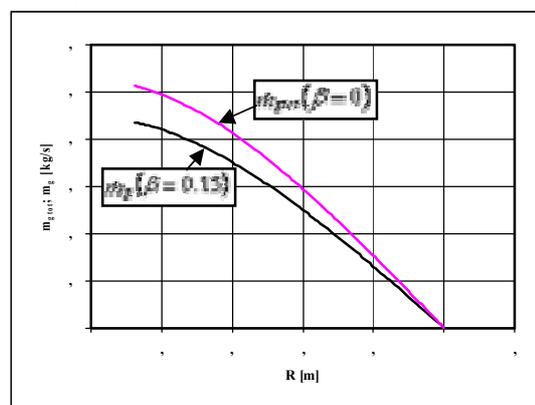


Fig 3 Gas mass flow distribution

2.1.2. Solid material flow towards the nozzle. Continuity equation written for particles

Using the symbols of Figure 2, the continuity equation for the particles in the control volume can be written in the following form:

$$R d\alpha dR \rho_b \varphi v_m + \left(R + \frac{dR}{2}\right) d\alpha b \left(\rho_m + \frac{d\rho_m}{2}\right) \left(v_m + \frac{dv_m}{2}\right) - \left[R - \frac{dR}{2}\right] d\alpha b \left(\rho_m - \frac{d\rho_m}{2}\right) \left(v_m - \frac{dv_m}{2}\right) - 0 \quad (11)$$

After transformation and rearranging, as well as neglecting second-order members, we obtain the following equation:

$$\frac{d\rho_m}{dR} = -\frac{\rho_b \varphi v_m}{b v_m} - \frac{\rho_m}{R} - \frac{\rho_m}{v_m} \frac{dv_m}{dR} \quad (12)$$

2.1.3. Momentum equation written for particles

The momentum equation can be written in the following form:

$$\left[\left(R + \frac{dR}{2}\right) \left(\rho_m + \frac{d\rho_m}{2}\right) \left(v_m + \frac{dv_m}{2}\right)^2 - \left(R - \frac{dR}{2}\right) \left(\rho_m - \frac{d\rho_m}{2}\right) \left(v_m - \frac{dv_m}{2}\right)^2 \right] b d\alpha = -dF + dF_f \quad (13)$$

„dF” on the right side of equation 13 means the drag force on the particles, while „dF_f” means the friction force.

These can be written in the following form:

$$dF = \frac{\rho_m}{m_1} b R d\alpha dR \frac{\rho_g}{2} A_o C_D \left(|v_g| - |v_m|\right)^2 \quad (14)$$

According to Kaskas [1], drag coefficient „C_D” can be written in the following form:

$$C_D = \frac{24}{Re} + \frac{4}{\sqrt{Re}} + 0.4$$

$$Re = \frac{(v_g - v_m) d_s \rho_g}{\mu_g} \quad (15)$$

The Coulomb friction force is:

$$dF_f = f \rho_m g R d\alpha b dR \quad (16)$$

After transformation and rearranging, as well as neglecting second-order members and taking equation 12 into consideration, we obtain the following relationship:

$$\frac{dv_m}{dR} = \frac{\rho_b \varphi v_m}{b \rho_m} - \frac{\rho_{go} P A_o C_D}{2 m_1 v_m p_o} \left(|v_g| - |v_m|\right)^2 + \frac{f g}{v_m} \quad (17)$$

The parameters of solid material-air mixture flowing towards the nozzle can be calculated with the use of equations 4, 10, 12, 17 and 19.

The calculations have shown that a dead zone is developed on the fluidization layer. This means that material flow does not start from the inside wall of the tank marked „R_t”, but only from the radius R_i < R_t. Therefore the value of radius „R_i” has to be high enough to satisfy condition F ≥ F_f for drag force „F” calculated with the radial air velocity determinable from gas mass flow „m_g” passing through surface section ΔA = π (R_t² - R_i²) of the distribution layer.

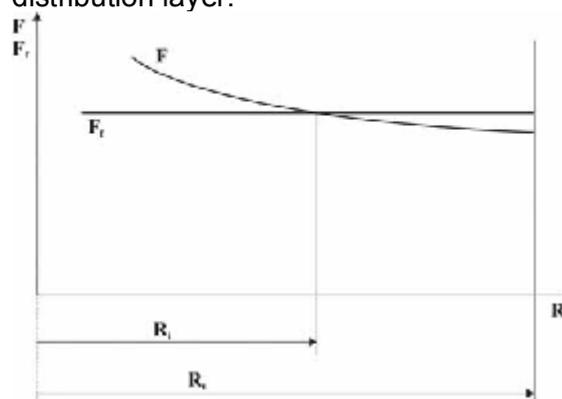


Fig. 4 Figure for determining „R_i” value

Thus, in order to start numerical calculations, the value of „R_i” has to be determined as the first step (see Figure 4).

Using equation 8, mass flow „m_g” is calculated at various „R” values.

The velocity of gas flowing towards the nozzle, at various values of „R” and with equation 4 taken into consideration, is obtained as:

$$v_g = \frac{\dot{m}_g}{2R\pi b \rho_g} = \frac{\dot{m}_g p_o}{2R\pi b \rho_{go} p} = \frac{\dot{m}_g p_o 2 a_n}{2R\pi b \rho_{go} (-b_n + \sqrt{b_n^2 + 4 a_n R})} \quad (18)$$

Drag force „F” is calculated by equation 14 with gas velocity „v_g” at particle velocity v_m=0. Repeating this at several assumed radii allows curve F(R) to be plotted. The Coulomb friction force “F_f” can be calculated by equation 16. The point of intersection of the two curves gives the „R_i” value (see Figure 4).

2.1.4. Diagrams obtained by solving the equations

Equations 4, 10, 12 and 17 have been solved using the numerical method of Runge-Kutta. The data of the example used are:

b=0.2 m; H=5 m; k_o=50000 m/s; m_m=30 kg/s; m_g=0.435 kg/s; p₁=145.3 kPa; p_f=139.23 kPa; p_k=133.23 kPa; R_f=1 m; R_p=0.125 m; v_k=100 m/s; β=0.15; ρ_b=800 kg/m³; ρ_s=2600 kg/m³; d_o≈1.5*10⁻⁴ m; solid material: fly ash; R_f=0.8074 m.

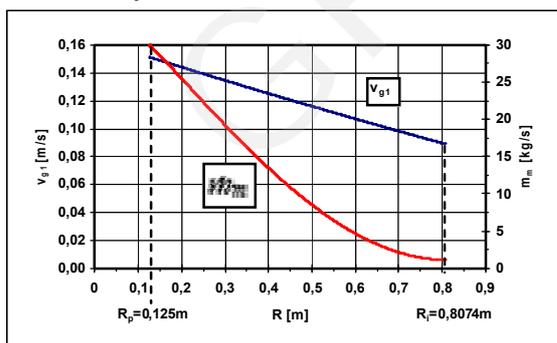


Fig. 5 Loosening velocity and solid material mass flow distribution

In Figure 5 the velocity distribution of air flowing through the distribution layer (layer 2 in Figure 1) and the mass flow variation flowing towards the nozzle are plotted as a function of radius.

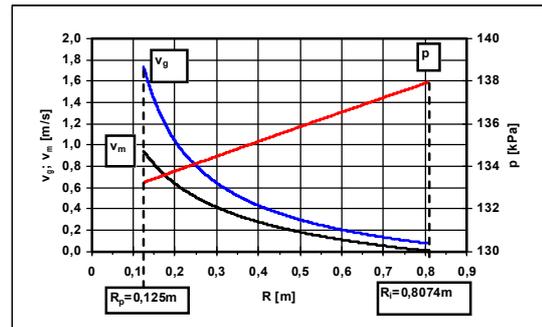


Fig 6 Gas-, solid material velocity and pressure distribution

Figure 6 shows the distributions of velocities in the gas flowing towards the nozzle, particle velocities and pressures over the distribution layer within the layer of „b” height (see Figure 2).

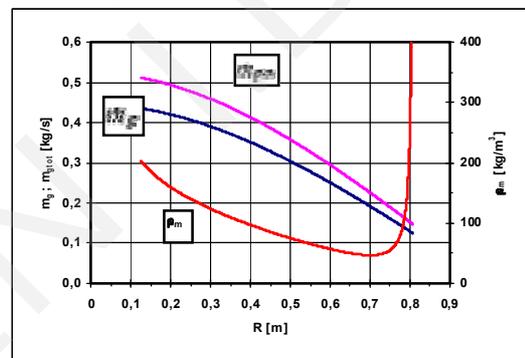


Fig. 7 Gas mass flow and material concentration distribution

Figure 7 shows the variations in material concentration „ρ_m” and radial gas mass flow „m_g”.

Conclusions and remarks

The model presented in this paper is suitable for taking the first step for the description of the complex flow taking place in air lifts. The equations derived, allows the main parameters to be calculated and the main dimensions of the equipment to be determined.

The following experiments need to be made in order to improve and verify the model used for the description of the two-phase flow taking place in air lift tanks.

- Determine relative number marked „φ” for velocity v_{mo} = φ v_m of material arriving from above into the control volume assumed in Section 2.1.

- The portion marked „ β ” of the air volume passing through the distribution layer provides the fluidization of the material column of „ H ” height. The value of „ β ” has to be determined by experiments.
- The pressure variation over the distribution layer was assumed to be parabolic in this paper. The value of this assumption has to be verified by experiments.
- The radius marked „ R_i ” of the dead zone discussed in Sub-section 2.1.2 has to be verified by experiments.

BHARAT ENGG/CALCUTTA also. He directed and controlled the putting into operation work of fly ash handling systems of power plants in Germany (Jänschwalde 1982), in Turkey (Kangal 1989-90 and 1999-2000), in Indonesia (Suralaya 1996-97 and 2002).

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Bio data kovacs@hds.bme.hu

Dr. L. KOVÁCS is a professor emeritus at the Budapest University of Technology and Economics, Department of Hydrodynamic Systems. He works more than 40 years in the field of pneumatic transport. He received in 1992 the degree of doctor from the Hungarian Academy of Sciences in recognition of his work in the theoretical development of pneumatic transport. He published more than 80 papers in the field of pneumatic transport.

Bio data varadi@hds.bme.hu

Dr. S. VÁRADI is an associate professor in the Budapest University of Technology and Economics, Department of Hydrodynamic Systems. His special field is to study and theoretical development of pneumatic transport. He is at present working as a private consultant in connection with the pneumatic transport in the EWB hungarian engineering concern, which is member of MCNALLY